

# 2 Design of computer experiments applied to modeling of compliant 3 mechanisms for real-time control

4 Diego A. Acosta · David Restrepo ·  
5 Sebastián Durango · Oscar E. Ruiz

6 Received: 27 October 2010 / Accepted: 19 June 2012  
7 © Springer-Verlag London Limited 2012

8 **Abstract** This article discusses the use of design of com-  
9 puter experiments (DOCE) (i.e., experiments run with a  
10 computer model to find how a set of inputs affects a set of  
11 outputs) to obtain a force–displacement meta-model (i.e., a  
12 mathematical equation that summarizes and aids in analyz-  
13 ing the input–output data of a DOCE) of compliant mecha-  
14 nisms (CMs). The procedure discussed produces a force–  
15 displacement meta-model, or closed analytic vector func-  
16 tion, that aims to control CMs in real-time. In our work, the  
17 factorial and space-filling DOCE meta-model of CMs is  
18 supported by finite element analysis (FEA). The protocol  
19 discussed is used to model the HexFlex mechanism func-  
20 tioning under quasi-static conditions. The HexFlex is a  
21 parallel CM for nano-manipulation that allows six degrees of  
22 freedom ( $x, y, z, \theta_x, \theta_y, \theta_z$ ) of its moving platform. In the  
23 multi-linear model fit of the HexFlex, the products or inter-  
24 actions proved to be negligible, yielding a linear model (i.e.,  
25 linear in the inputs) for the operating range. The accuracy of  
26 the meta-model was calculated by conducting a set of com-  
27 puter experiments with random uniform distribution of the  
28 input forces. Three error criteria were recorded comparing  
29 the meta-model prediction with respect to the results of the  
30 FEA experiments by determining: (1) maximum of the

absolute value of the error, (2) relative error, and (3) root  
mean square error. The maximum errors of our model are  
lower than high-precision manufacturing tolerances and are  
also lower than those reported by other researchers who have  
tried to fit meta-models to the HexFlex mechanism.

**Keywords** Design of computer experiments · Design of  
experiments · Compliant mechanism · Meta-modeling ·  
Plackett-burman design · Uniform design

<b>List of symbols</b>	40	
XYZ	Fixed reference coordinate system	41
T1	Input force port on Tab1	42
T2	Input force port on Tab2	43
T3	Input force port on Tab3	44
D1	Direction parallel to the connection beams in the HexFlex	45
D2	Direction perpendicular to the plane that contains the HexFlex on its relaxed configuration	46
$\tau$	Vector of input forces and torques	47
$\mathbf{r}$	Configuration of the end effector	48
$x, y, z$	Coordinates of a point in XYZ frame	49
$\theta_x, \theta_y, \theta_z$	Set of XYZ Euler angles	50
DOE	Design of experiments	51
DOCE	Design of computer experiments	52
CM	Compliant mechanism	53
DOF	Degrees of freedom	54
FEA	Finite element analysis	55
PRBM	Pseudo-rigid body modeling	56

A1 D. A. Acosta (✉)  
A2 Grupo de Investigación DDP, Universidad EAFIT,  
A3 Calle 49 No 7–sur–50, Medellín, Colombia  
A4 e-mail: dacostam@eafit.edu.co

A5 D. Restrepo · O. E. Ruiz  
A6 Laboratorio de CAD CAM CAE, Universidad EAFIT,  
A7 Calle 49 No 7–sur–50, Medellín, Colombia

A8 S. Durango  
A9 Grupo de Investigación DM-DI, Universidad Autónoma de  
A10 Manizales. Antigua estación del Ferrocarril,  
A11 Manizales, Colombia

## 1 Introduction 59

In traditional mechanisms, movement is achieved using  
kinematic joints (cylindrical, spherical, prismatic, etc.) and 61

links that are as rigid as possible. In contrast, compliant mechanisms (CMs, [1]) are mechanical devices which undergo elastic deformations to transmit motion, force or energy from specified input ports to output ports. The main advantage of CMs with respect to traditional rigid-link mechanisms is that fewer parts and assembly processes, and no lubrication is required.

Due to the complexity of their motion (which is actually a deformation), CMs cannot be designed and directly analyzed by traditional kinematic methods [2]. Computational methods which require fast appraisal of the CM response to forces or, conversely, forces needed to bring the CM to a given configuration are needed for real-time control of the CM. Those methods usually imply the availability of a closed analytical (input/output, I/O) function, which can be computed directly and inversely. As a result of our work, the closed-form analytical function was found and, in addition, it is invertible around the operating point.

Two methods are used to relate CM deformations against forces and/or torques [3]: (1) pseudo-rigid body modeling (PRBM) and (2) numerical methods such as shooting methods (SM), finite element methods (FEM) and the chain algorithm [4]. In PRBM, the rigid body analysis is extended to the flexible-body analysis by finding the deflection of a flexible link by approximating closed-form functions [4, 5]. This aspect considerably limits the applicability of PRBM as the approximated model is usually not accurate enough for precision applications [5]. On the other hand, if a numerical method is used to model CMs, the designer is basically solving a continuum mechanics problem by a discrete (computational) strategy. Therefore, the accuracy of the results critically depends on the resolution of the discretization. Since each run of the numerical solution requires considerable numerical processing, these methods are not suitable for time-critical applications (e.g., real-time control).

The term meta-model refers to an approximated I/O function to fit the I/O data produced by computer simulations of a model. Meta-modeling has been used to model a variety of complex systems. Reference [6] presents the application of meta-modeling on helicopter tests. Application of meta-modeling for vehicle testing analysis is presented in [7]. More general examples of applications of meta-modeling appear in Ref. [8, 9].

This article presents a design of computer experiments (DOCE) methodology to generate meta-models of CMs that synthesize a force (input)–displacement (output) model of CMs working under quasi-static conditions. In general, the lack of tools to model and analyze CMs is recognized as an open research problem [5], and these tools are actually required for real-time control. This methodology presents the advantages of DOCE that can be used to fit meta-models of any type of CMs.

The structure of this article is as follows: Sect. 2 presents a literature review and contrasts the contributions, Sect. 3 presents the proposed methodology and its scope for force–displacement modeling of CMs under quasi-static conditions, the case study “HexFlex” CM is developed in Sect. 4, where the proposed methodology is applied to obtain a mathematical meta-model that relates the actuator forces at the input ports with the end effector configuration. The mechanism, input factors (input forces), and their levels are described in Sect. 4.1. Section 4.3 develops the Fractional DOCE to determine the main factors. Section 4.4 presents the space filling DOCE and the meta-modeling of the HexFlex CM by conducting FEA tests. Section 4.6 successfully validates the meta-model just obtained by running 1,000 FEA tests and comparing their results against the meta-model predictions. Sections 5 and 6 conclude the article.

## 2 Literature review, modeling of compliant mechanisms

### 2.1 Meta-models

The term *meta-model* in computer experiments represents a surrogate model based on the use of statistical techniques to yield mathematical equations that approximate the results rendered by computer algorithms such as FEA [10]. If the true nature of a computer analysis code is  $\mathbf{u} = \mathbf{u}(\mathbf{v})$  with  $\mathbf{v}$  being the input variables vector and  $\mathbf{u}$  the output variables vector from the computer code, then a surrogate model (i.e., meta-model) of the computer analysis is  $\hat{\mathbf{u}} = \mathbf{z}(\mathbf{v})$  with  $\hat{\mathbf{u}}$  being an approximation of the output variables defined by a functional relation  $\mathbf{z}(\mathbf{v})$  found statistically. This carries an approximation error or residual defined as  $\epsilon = \hat{\mathbf{u}} - \mathbf{u}$ .

Meta-models have benefits in screening variables, reducing design costs and optimizing designs [11]. They are applied here to model the quasi-static behavior of the HexFlex mechanism. The HexFlex is a six degrees of freedom parallel CM with distributed compliance for nanomanipulation designed at the MIT by Martin L. Culpepper and Gordon Anderson [12, 13].

### 2.2 Force–displacement modeling of compliant mechanisms

Topology and geometry optimization methods applied to CMs allow to tune up shapes, dimensions and connectivities to achieve a good numerical value of a function which evaluates the efficiency of the CM. The optimization relies on the possibility of relating forces/torques versus positions/deformations in the mechanism [3]. For such purpose,

162 structural optimization uses either PRBM or numerically  
 163 solved differential equations (FEA, SM, chain algorithms)  
 164 or a conceptual synthesis tool [14] in which a building  
 165 block approach is used to obtain a feasible initial design of  
 166 known size and geometry and fine tuned using optimization  
 167 methods. This literature review is focused on the analytical  
 168 and numerical methods used as modeling tools in the  
 169 analysis and design of CMs. A comparison of accuracy,  
 170 computational efficiency and usability of methods for large  
 171 deflection analysis of a cantilever beam (a specific CM  
 172 member category) under free-end point load cases has been  
 173 reported in Ref. [15], whose discussion and conclusions  
 174 about the accuracy and computational efficiency of the  
 175 studied methods are agree with our survey.

### 176 2.3 Pseudo-rigid body modeling

177 Pseudo-rigid body modeling is used to design CMs by  
 178 representating compliant members by rigid link equivalents  
 179 paired with standard joints (prismatic and revolute) and  
 180 coupled with springs of appropriate stiffness [3, 14]. In this  
 181 way, the CM behaves as a mechanism (i.e., has degrees of  
 182 freedom) although strictly speaking no kinematic joints are  
 183 present.

184 In the designed device, the PRBM differentiates stiff  
 185 from flexible components. The first ones are modeled as  
 186 completely rigid while the latter ones provide mobility to  
 187 the mechanism (links coupled with displacement and torsional  
 188 springs, non-linear elastic beams, etc.). These hyper-  
 189 flexible members can be analyzed with closed differential  
 190 equations (e.g., flexural cantilever beam). A key step of the  
 191 PRBM is to estimate the equivalent application point and  
 192 equivalent elastic constant of the springs representing the  
 193 compliant elements, i.e., the topology, geometry and elastic  
 194 characteristics of the equivalent mechanism.

195 The PRBM approach is mathematically addressed under  
 196 linear and non-linear strain formulations. This means that  
 197 the strains are expressed in terms of linear and non-linear  
 198 displacements. From elasticity theory, strains can be for-  
 199 mulated as functions of the partial derivatives of the dis-  
 200 placement functions, and, usually, higher-order partial  
 201 derivatives.

202 The linear formulation neglects partial derivatives that  
 203 have an order or power larger than one. The following  
 204 articles present linear PRBM as a fundamental part of their  
 205 formulation: analytical models of revolute and translational  
 206 compliant joints are presented in Ref. [16]. In Ref. [17],  
 207 PRBM is applied in predicting the behavior of a nano-scale  
 208 parallel guiding mechanism which uses two carbon nano-  
 209 tubes as flexural links. The kinematic behavior was reported  
 210 to be 92.7 % accurate with respect to a molecular simula-  
 211 tion. In Ref. [18], the kinematic and force analysis of  
 212 compliant-driven robotic mechanisms is based on equations

213 that relate joint torques, joint angles and displacements. In  
 214 Ref. [19], the I/O model of a compliant micro-motion stage  
 215 equivalent to a parallel mechanism formed by three limbs  
 216 with rotational–rotational–rotational (3RRR) topology is  
 217 obtained replacing the flexures with equivalent springs.

218 Non-linear PRBM is based mainly on the application of  
 219 Euler beam models or deflection models based on the  
 220 Castigliano's second theorem to model the flexible mem-  
 221 bers of the CM solving high order partial derivatives of the  
 222 strain formulation. The following articles present non-lin-  
 223 ear PRBM as part of their formulation: Ref. [20] discusses  
 224 conic section flexure hinges using Euler beam model and  
 225 Castigliano's second theorem. Reference [21] introduces  
 226 an analytical approach to corner filleted flexure hinges  
 227 using the Castigliano's second theorem. Reference [22]  
 228 develops a synthesis and analysis PRBM for the limit  
 229 configurations of a four-bar mechanism with an output  
 230 compliant link (one end pinned to the coupler, one end  
 231 fixed to the ground). The lumped compliance is modeled  
 232 by non-linear beam theory, allowing for large non-linear  
 233 deflections of the pinned end of the compliant link. The  
 234 model only applies for a given topology. In Ref. [5], PRBM  
 235 is enhanced to allow large deflections of elastic hinges.  
 236 Four elastic hinges (leaf spring, cross, notch, and Haber-  
 237 land) are modeled and a joint-based modular approach is  
 238 obtained. The modeling technique reported reduces the  
 239 time needed for off-line modeling and design but not  
 240 enough for real-time control. Reference [23] presents the  
 241 mathematical model derived from the second Castigliano's  
 242 theorem, for a six degrees of freedom (DOF) CM. The  
 243 forward and inverse analyses of an open loop CM are  
 244 developed in Ref. [24] using numerical methods to solve  
 245 large deformations of the mechanism. Reference [25]  
 246 develops a mathematical dynamic model, based on large-  
 247 deflection beam models, for compliant constant force  
 248 compression mechanisms. In Ref. [26], a large deflection  
 249 analysis of compliant beams is presented. The method is  
 250 based on the Adomian decomposition method in which  
 251 differential equations are solved by a semi-analytical  
 252 strategy different from the Euler beam or Castigliano's  
 253 second theorem formulations. The method is reported to be  
 254 efficient and accurate with respect to numerical and linear  
 255 solutions. However, it is exclusively formulated for canti-  
 256 lever-like compliant members.

257 A model obtained with linear PRBM can be usually  
 258 applied in real-time control but is restricted in precision  
 259 engineering applications because of its low accuracy [1, 5].  
 260 Non-linear PRBM is suitable for accurate modeling and  
 261 design, but it is not computationally efficient for real-time  
 262 control. At any rate, PRBM requires that the geometry and  
 263 loads of the elastic links allow for a closed-form analytic  
 264 solution. These considerations seriously hinder the appli-  
 265 cation of PRBM.

## 266 2.4 Numerical methods

267 When the CM do not accept closed analytical force/  
268 deformation solutions, and computing time is not an issue,  
269 numerical methods are applicable for their analysis and  
270 simulation. Reference [27] presents a procedure for the  
271 optimal design of flexural hinges for compliant micro-  
272 mechanisms. The optimal design is developed by coupling  
273 a FEA model to an optimization algorithm. The optimi-  
274 zation is intended to maximize the rotation of the hinges  
275 under kinematic and strain constraints of the material of the  
276 hinge. Because of its time expenses, a pure FEA modeling  
277 of CMs is restricted to the design stage of the mechanism,  
278 being excluded from real-time control applications.

279 Reference [28] presents a localized application of FEA  
280 in CMs. In the design stage, two main steps are taken to  
281 complete a force–deformation model: (1) The elastic  
282 properties of the hinges are estimated by an independent  
283 3D FEA. (2) The FEA-estimated properties are incorpo-  
284 rated into a general CM model by the use of equivalent  
285 beams. This hybrid model may be applied in a reduced  
286 manner for real-time I/O models of CMs. The limitation  
287 exists because the geometry of the zones in which the  
288 equivalent flexible beam meets the rigid parts has a con-  
289 siderable influence on the predictions of deformations and  
290 stress concentrations of the CM.

291 Reference [29] presents the synthesis of CMs. The merit  
292 of the article is that, unlike others, it extensively presents  
293 the usage of highly non-linear finite elements, allowing the  
294 modeling of very large deformations.

295 In Ref. [30], the stiffness properties of a (compliant)  
296 notch hinge are computed using FEA relating the initial  
297 and final mechanism configurations under known loads.  
298 The procedure is only used to find the properties of the  
299 flexures and not to find an I/O model of the CM.

300 Reference [4] presents the use of a generalized shooting  
301 method (GSM) for the case of CMs with curved members. The  
302 method preserves the computational advantages of SM over  
303 finite differences and FEM: boundary value problems are  
304 treated as initial value problems instead of relying on fine  
305 discretization of the beam members to achieve high accuracy.

306 Summarizing, numerical methods such as FEM or SM are  
307 useful in determining the deflection and stresses in CMs  
308 because they allow to analyze CMs that have a geometry that  
309 is not easily modeled using methods like the PRBM. However,  
310 numerical methods cannot be used in a real-time scenario to  
311 control CMs. For this purpose, an intermediate I/O model  
312 must be estimated. This is the purpose of this investigation.

## 313 2.5 Contribution of this article

314 This article presents a new general procedure for modeling  
315 CMs under quasi-static conditions by DOCE methodology.

The proposed approach allows the modeling of CMs that 316  
have lumped or distributed compliance with simple or 317  
complex geometry. The main advantages of the proposed 318  
approach with respect to traditional modeling methods 319  
(PRBM, FEM, SM, chain algorithms) are: 320

1. The methodology is general enough to cover both 321  
lumped and distributed CMs. 322
2. The obtained input–output model might be simple 323  
enough to be used in real-time control. 324
3. Real experimentation is replaced by computer simu- 325  
lations reducing costs in product development. 326

It is clear that DOCE does not replace design of 327  
experiments (DOE) (i.e., physically conducted experi- 328  
ments). However, the pre-fitting of the model using DOCE 329  
serves to identify and avoid ranges, interactions and limi- 330  
tations that would make the DOE extremely expensive. As 331  
an application of the methodology, the 6 DOF CM HexFlex 332  
is modeled by finding an accurate model with respect to 333  
FEA simulations. 334

The differences in application domains between PRBM, 335  
numerical methods and DOCE techniques should be 336  
remarked here: PRBM is only applicable to mechanisms 337  
whose geometry can be decomposed into links for which 338  
an closed analytical expression for force versus deforma- 339  
tions is possible (e.g. uniformly extruded beams). Numeri- 340  
cal methods as FEM present no restrictions in the 341  
geometry of the mechanisms or bodies being analyzed. 342  
However, it is a slow method definitely not suited for real- 343  
time applications. DOCE allows to calibrate systems (not 344  
only mechanisms) and to obtain an I/O model which is fast 345  
and accurate for the chosen operation point. In our case, 346  
DOCE uses FEA as a subsidiary tool to carry the computer 347  
experiments and therefore allows to tune up the I/O model. 348  
In DOCE, if the mechanism or the operating point is 349  
changed, a new DOCE model is required. In a typical 350  
application of DOCE for mechanisms, a given mechanism 351  
is calibrated or modeled via computer experiments. Next, it 352  
is installed in the host device and then controlled via the 353  
analytical I/O model previously obtained. 354

### 3 Methodology for meta-modeling of compliant mechanisms 355 356

#### 3.1 Design of computer experiments 357

In mechanism and machine science defining the configu- 358  
ration of a mechanism is equivalent to determining the 359  
positions of all moving points, or the location of all bodies, 360  
or specifying all joint parameters, etc. In this sense, 361  
defining the spatial configuration of a body (as the end 362  
effector of a CM) is equivalent to determining six 363

364 parameters, e.g. the three coordinates of a point and a set of  
 365 three Euler angles. In meta-modeling of CMs, we are  
 366 seeking a function  $f$  that relates the input forces and torques  
 367 ( $\boldsymbol{\tau}$ ) with the configuration ( $\mathbf{r}$ ) of the end effector:

$$f : \boldsymbol{\tau} \rightarrow \mathbf{r}$$

$$\boldsymbol{\tau} = [\tau_1 \tau_2 \cdots \tau_n]^T \quad (1)$$

$$\mathbf{r} = [r_1 r_2 \cdots r_m]^T$$

369 with  $m \leq n$ .

370 A redundant mechanism occurs when  $m < n$ , while a  
 371 non-redundant one occurs when  $m = n$ . In our discussion,  
 372 we assume that the addressed mechanisms are not redun-  
 373 dant. We also recall that for an end effector taking an  
 374 spatial configuration, we have  $m = 6$ . The methodology  
 375 presented in Fig. 1 is proposed to model CMs under quasi-  
 376 static conditions using meta-models from computational  
 377 experiments. This methodology is based on results  
 378 obtained by numerical simulations of CMs using FEA and  
 379 is summarized as follows:

- 380 1. Define the topology and geometry of the CM. The  
 381 topology concerns the connectivity and joint types of  
 382 the CM. The geometry addresses the initial configu-  
 383 ration as well as the exact shapes and sizes of the CM.
- 384 2. Define the set of factor parameters. That is, the vector  
 385 of input forces ( $\boldsymbol{\tau}$ ).
- 386 3. Propose a meta-model of the CM. This model is  
 387 usually a multi-input multi-output (MIMO) poly-  
 388 nomial model, calculated in the vicinity of the operat-  
 389 ing point. In Ref. [31], Rao states that MIMO regres-  
 390 sion models are an extension of SISO ones, whenever they  
 391 have the same experimental design. Ordinary least  
 392 squares (OLS) can be used for each individual output  
 393 in independent manner. This usage has as a pre-  
 394 condition that the outputs be actually independent  
 395 from each other, while dependent on the input  
 396 variables.
- 397 4. Use a fractional factorial DOCE (e.g., Plackett–Bur-  
 398 man) to screen variables. The main factors are  
 399 obtained by computer experiments. Fractional factorial  
 400 DOEs can be used to screen  $n$  factors when the number  
 401 of runs is a power of 2 (i.e., 4, 8, 16, 32, . . .). The gaps  
 402 between these numbers widen considerably as  $n$   
 403 increases [37]. Plackett and Burman devised a frac-  
 404 tional factorial DOE in which the number of runs is a  
 405 multiple of four for any number  $n$  of factors (i.e., 12,  
 406 20, 24, 28, . . .) [32].
- 407 5. Use a space filling DOCE such as uniform design [33]  
 408 to fine-tune the mathematical model of the mechanism  
 409 by computer experiments.
- 410 6. Construct the surrogate model of the kinematics of the  
 411 CM.

7. Verify the accuracy of the meta-model using extra  
 experiments [10].

The previous steps define the scope of our article.  
 However, it must be kept in mind that actual experiments  
 must be conducted to fit an industrially applicable meta-  
 model. Our contribution permits to execute these experi-  
 ments with a minimum overhead cost.

### 3.2 Methodology scope

The presented methodology for modeling CMs is limited  
 to:

1. CMs that allow small displacements of its end effector.
2. Input forces and moments slowly varying in time (quasi-static conditions).
3. The model is restricted to the neighborhood of the operation point for which was calculated.
4. Although the proposed methodology is general for CMs, the obtained force–displacement models are specific for each individual case.

In spite of these limitations, the proposed force–dis-  
 placement modeling of CMs by DOCE is relevant for a  
 wide range of applications because most CMs are designed  
 for small displacements of its end effector under quasi-  
 static conditions, specially in compliant parallel nano-  
 manipulating mechanisms. Because of this, they can be  
 modeled by a low order Taylor series polynomial.

It should be emphasized that DOCE methodology is  
 suitable when a empirical input–output model is required  
 for a phenomenon or system which are fundamentally

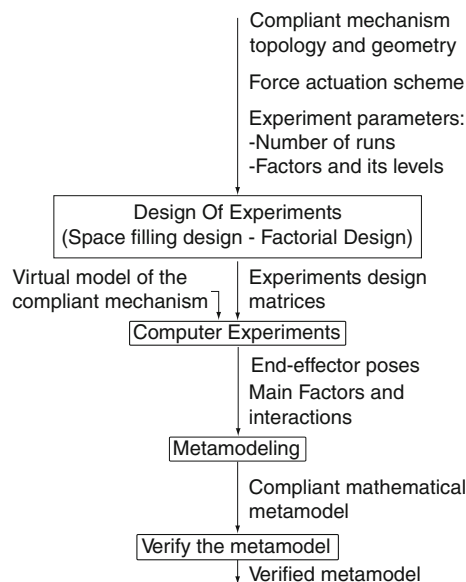


Fig. 1 Methodology for force–displacement meta-modeling of compliant mechanisms

Author Proof

441 difficult to model theoretically. We fit a DOCE model  
 442 which relates forces (inputs) to end effector configuration  
 443 (outputs) of a mechanism. The application scenario of such  
 444 a model is the real-time control of the mechanism in the  
 445 vicinity of an operating point. Because one is interested in  
 446 real-time control, an accurate but slow FEA model is out of  
 447 question. The DOCE model might be a linear, quadratic,  
 448 cubic, etc. approximation around the operating point.  
 449 DOCE allows to trim the polynomial model by neglecting  
 450 high level interactions if their statistical significance is low.  
 451 In this article, this trimming led to a linear meta-model.

452 Section 4 shows the meta-modeling of the HexFlex CM  
 453 under quasi-static conditions.

#### 454 4 Case study: force–displacement meta-modeling 455 of the HexFlex mechanism

456 Applying the procedure described in Sect. 3, the HexFlex  
 457 parallel CM is meta-modeled.

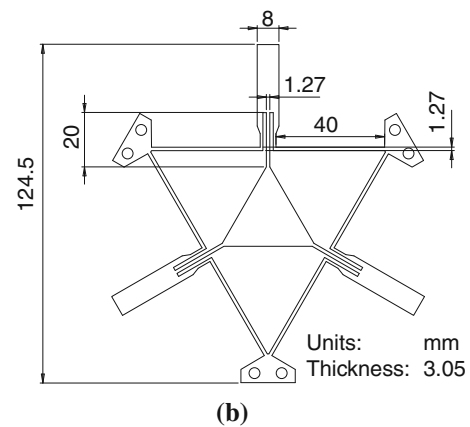
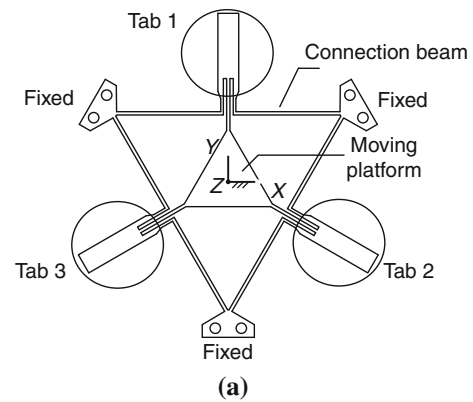
##### 458 4.1 The HexFlex mechanism

459 The topology and dimensions of the HexFlex are shown in  
 460 Fig. 2. The HexFlex composed of a triangular moving  
 461 platform, three tabs to provide an interface with the actuators,  
 462 and six connection beams between the moving  
 463 platform and the grounded zone (Fig. 2a). This mechanism  
 464 allows the spatial motion of the moving platform, then, the  
 465 end effector configuration is specified by six independent  
 466 movements (e.g. translation and rotation, on the X, Y and  
 467 Z axes) as shown in Fig. 3.

468 To control the moving platform, there are two actuators  
 469 in the external edge of each tab. For each tab, one actuator  
 470 acts in direction parallel to the connection beams (called  
 471 direction one and denoted D1) and, the other actuator acts  
 472 perpendicular to the tab (Z direction, D2) as in Fig. 4. Tabs  
 473 are denoted T1, T2, T3. The motion of an specific actuator  
 474 is denoted by the tab followed by the direction using the  
 475 convention shown in Fig. 4.

476 The actuators used in the experiments allow a force of 1 N.  
 477 The positive direction of actuators for D2 coincides with the  
 478 direction in which Z is positive, and for D1 the positive  
 479 direction of actuators is as shown in Fig. 4. Forces which vary  
 480 slowly with time are assumed for the experiments (quasi-static  
 481 experiments). Planar and non-planar displacements can be  
 482 simultaneously achieved, by driving the tabs inside the planes  
 483  $\Pi$  shown as an example in Tab 3 in Fig 3e. The material  
 484 selected to model the mechanism is Aluminum 7075.

485 To define the meta-model function, the vector of input  
 486 forces ( $\tau$ ) and end effector configuration ( $\mathbf{r}$ ) are defined by:



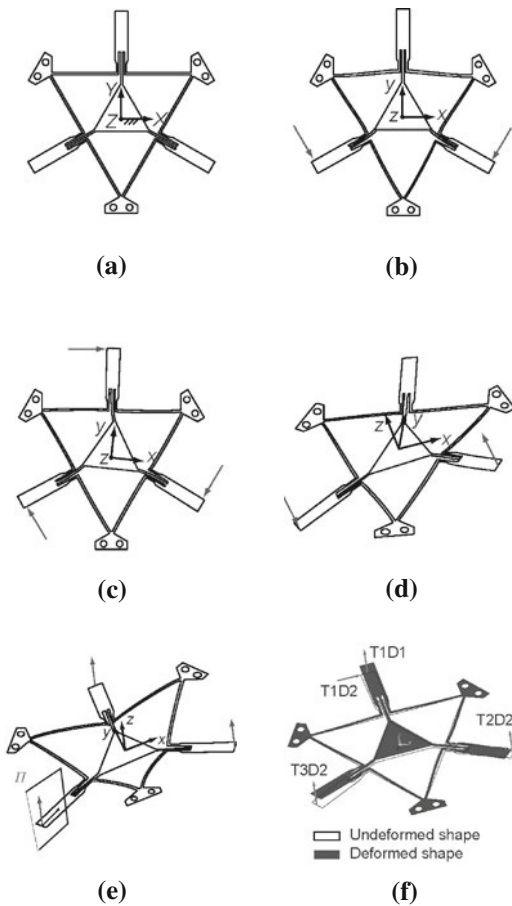
**Fig. 2** Six DOFs compliant mechanism [34]. **a** HexFlex components. **b** HexFlex main dimensions

$$\tau = [T1D1 \quad T1D2 \quad T2D1 \quad T2D2 \quad T3D1 \quad T3D2]^T \quad (2)$$

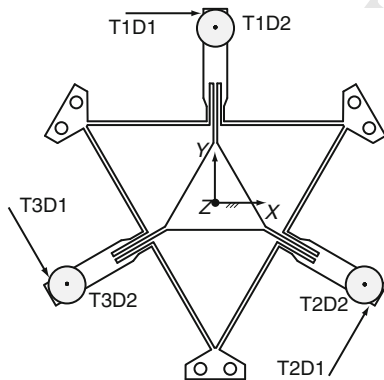
$$\mathbf{r} = [x \quad y \quad z \quad \theta_x \quad \theta_y \quad \theta_z]^T \quad (3) \quad 488$$

490 where the end effector configuration is defined by six  
 491 independent parameters: the coordinates ( $x, y, z$ ) of the  
 492 moving platform center and a set ( $\theta_x, \theta_y, \theta_z$ ) of XYZ small  
 493 Euler angles. Euler angles represent the final orientation of  
 494 the moving stage as a succession of three rotations that take  
 495 place around an axis whose location depends upon the  
 496 preceding rotations [35]. However, in the case of small  
 497 Euler angles, the order of the X, Y and Z rotations do not  
 498 affect the represented orientation [36]. The input forces  
 499 ( $T_i D_j$ ) correspond to the actuators in the tabs (tab  $i$ , force  
 500 direction  $j$ ).

501 The reference frame or world coordinate system is  
 502 chosen to be at the center of the moving platform in its  
 503 relaxed configuration (Fig. 3a). The moving platform  
 504 coordinate frame is attached at its centroid. Therefore, in  
 505 the relaxed configuration, the reference frame and the  
 506 moving frame coincide.



**Fig. 3** Six DOFs compliant mechanism in deformation. **a** Relaxed position, **b** in-plane translations, **c** in-plane rotations, **d** out-plane rotations, **e** out-plane translations and **f** deformed shape



**Fig. 4** HexFlex actuators direction

507 4.2 Workflow of FE experiments and used software  
508 packages

509 Figure 5 displays the assembly of computational tools to fit  
510 the DOCE model of CMs. This model is a set of hyper-  
511 surfaces ( $r_1, r_2$ , etc.) in the multidimensional space of the

input variables ( $\tau_1, \tau_2$ , etc.), as shown in the upper right  
512 corner. The statistical fitting of the model (upper left corner  
513 of Fig. 5) is carried out with MATLAB<sup>®</sup> code, using a  
514 Plackett–Burman DOE. Such values are calculated by a  
515 number of FEA simulations in ANSYS<sup>®</sup> (lower row of  
516 Fig. 5) controlled from the MATLAB<sup>®</sup> programs.  
517 ANSYS<sup>®</sup>, therefore, acts in this case as a FEA server  
518 subordinated to MATLAB<sup>®</sup> programs.  
519

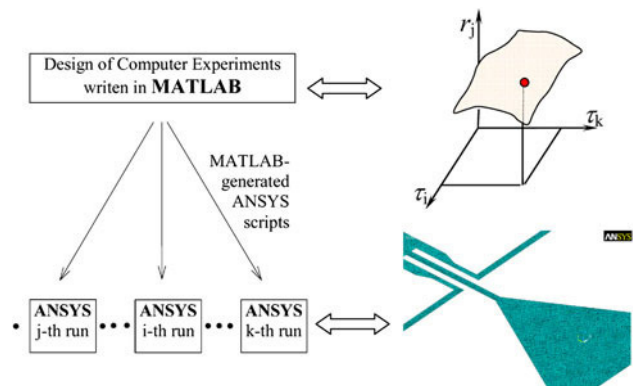
520 Using the symmetry of the mechanism and the dimen-  
521 sions shown on Fig. 2b, a sixth part of the mechanism was  
522 modeled and meshed to make a geometric FEM model of  
523 the mechanism (Fig. 6a). Using geometric transformations,  
524 the mechanism was completed developing a symmetrical  
525 mesh. Then the mesh was exported to ANSYS<sup>®</sup> using quad  
526 shell elements to run the DOCE (Fig. 6b). The computer  
527 experiments obtained the moving frame configuration  
528 given a set of input loads on the tabs.

529 *Operation ranges* The selection of operation ranges for  
530 the DOCE is based on the recommendations of the Hex-  
531 Flex mechanism designers [34], corresponding to a force  
532 range of  $\pm 1$  N. As for the operating point for the HexFlex  
533 mechanism, we adopted the usual one in the literature,  
534 which is  $x = 0, y = 0, z = 0, \theta_x = 0, \theta_y = 0$ , and  $\theta_z = 0$ .  
535 It must be pointed out that, in addition to the reviewed  
536 literature, the ranges, convenience of the operating point,  
537 and elimination of high degree and crossed-influence terms  
538 were verified by the series of computer experiments carried  
539 out by the methodology applied in this article.

540 The upper and lower levels of each factor for the frac-  
541 tional factorial and space filling DOCE are displayed on  
542 Table 1. The factors or inputs of the experiments are  
543 defined by Eq. (2) and correspond to the actuation forces  
544 of the mechanism, which are the controllable input variables  
545 (i.e., factors) of the experiment.

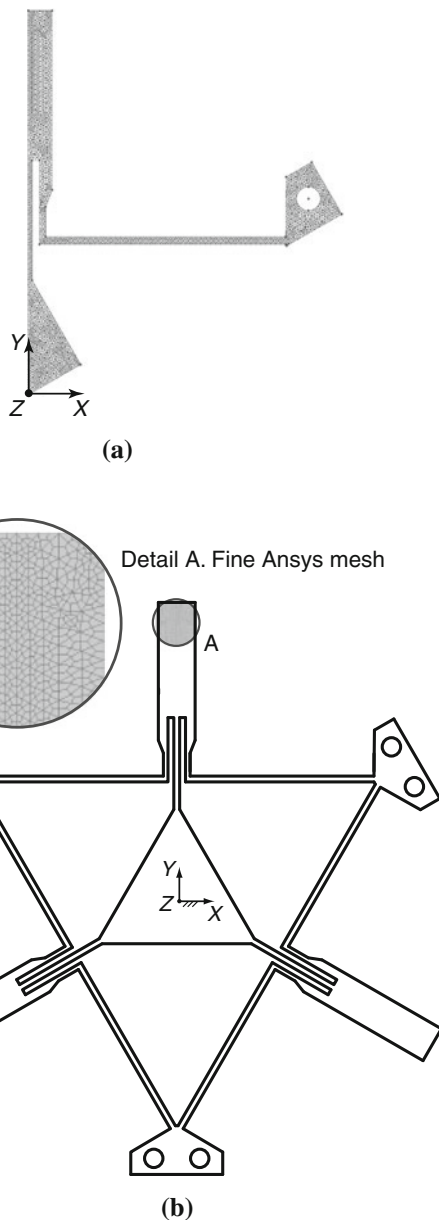
4.3 Fractional factorial DOCE

546 The statistical fitting of the model (upper left corner of  
547 Fig. 5) is carried out with MATLAB<sup>®</sup> code, using a  
548



**Fig. 5** ANSYS<sup>®</sup> as a FEA server controlled from MATLAB<sup>®</sup>

Author Proof



**Fig. 6** FEA model of the HexFlex Mechanism. **a** Sixth part of the mesh. **b** FEM model in ANSYS<sup>®</sup>

fractional factorial DOCE. For this purpose, a Plackett–Burman fractional factorial DOCE was chosen in lieu of a  $2^{6-3}$  despite its higher number of runs (namely, 12) because of its higher resolution [37]. To screen factors, a Plackett–Burman (PB) DOCE [37–39] with 12 runs is made. Plackett–Burman designs are very economical and efficient when only main effects are of interest.

Provided that interactions are negligible (a hypothesis verified later using a uniform design of experiment), PB DOCE can be analysed using Lenth and Daniel plots [37]. Other accepted methods of analysis are the mean squared

**Table 1** Studied factors

Factor	Low level (N)	High level (N)
T1D1	−1	+1
T1D2	−1	+1
T2D1	−1	+1
T2D2	−1	+1
T3D1	−1	+1
T3D2	−1	+1

Forces in Tabs of the HexFlex

residuals and Bayes discrimination model determination [37]. In this paper, Lenth and Daniel plots are used to analyze PB DOCE and the assumption of negligible interactions is verified using a uniform DOCE. A script was developed to automatically generate the computer experiments and their results. The DOCE matrix and the results of each response are shown in Table 2.

To analyze the results of the PB DOCE, half normal probability (HNP) (Fig. 7) and Pareto (Fig. 8) plots are calculated. As an example, Fig. 8a means that  $x \approx (11.5 \times T1D1 + 6 \times T2D1 + 6 \times T3D1) \times 10^4$ . These analyses provide a simple way to examine the response variables (i.e.,  $x$ ,  $y$ ,  $z$ , etc) and the relative importance of the factors of the experiment. The Pareto plots results coincide with half normal probability (HNP) showing that the main effects are consequent with the symmetries of the topology of the mechanism (Fig. 4). The symmetries of the mechanism also made that some effects had the same value.

In addition to the Pareto plots and HNP analysis, another way of looking at the resulting effects consists of using Lenth's plot [40]. The absolute values of the alias of the effects (i.e., estimate of an effect also including the influence of one or more other effects (usually high order interactions) in a fractional factorial DOE [38]) are ordered in ascending order to calculate the median ( $v$ ). Once the median is calculated a pseudo-standard error ( $S_0$ ) is estimated using the formula:  $S_0 = 1.5 v$ . The pseudo-standard error serves to define the margin of error (ME) and the simultaneous margin of error, using the 0.975-quantile and  $t_{g,m/3}$  of the  $t$  student distribution allowing fractional degrees of freedom. The results for these analyses are displayed on Table 3 and indicate which independent variables (T1D1, T1D2, T2D2, T3D2, T2D1, T3D1) have effects on which dependent variables ( $x$ ,  $y$ ,  $z$ ,  $\theta_x$ ,  $\theta_y$ ,  $\theta_z$ ). Because this is a fractional factorial design, we can only screen at this point the existence of such dependency. Later on, using a space filling technique we will confirm and quantify them. The responses are affected as follows:

1.  $x$  and  $\theta_z$  are mainly affected by T1D1, T2D1, and T3D1.



**Table 2** Plackett–Burman DOCE matrix for 6 factors and 12 runs

Design matrix						Responses					
T1D1 (N)	T1D2 (N)	T2D1 (N)	T2D2 (N)	T3D1 (N)	T3D2 (N)	$x$ ( $\mu\text{m}$ )	$y$ ( $\mu\text{m}$ )	$z$ ( $\mu\text{m}$ )	$\theta_x$ ( $\mu\text{rad}$ )	$\theta_y$ ( $\mu\text{rad}$ )	$\theta_z$ ( $\mu\text{rad}$ )
1	-1	1	-1	-1	-1	115,056	0.6	-862,976	0.0001	-0.0001	3.10176
-1	-1	1	1	1	-1	-57,529	99,636.5	-287,659	-39.3596	-68.0656	3.10183
1	-1	-1	-1	1	1	-3	-0.6	-287,655	-39.2665	68.1194	-9.30545
-1	-1	-1	-1	-1	-1	-57,525	-99,636.5	-862,976	0.0001	-0.0001	3.10186
1	1	-1	1	1	-1	-3	-0.6	287,655	39.2665	-68.1194	-9.30545
1	1	1	-1	1	1	57,525	99,636.5	287,659	39.3596	68.0656	-3.10186
-1	-1	-1	1	1	1	-115,056	-0.6	287,662	-78.6262	0.0539	-3.10176
-1	1	-1	-1	-1	1	-57,525	-99,636.5	287,659	39.3596	68.0656	3.10186
-1	1	1	1	-1	1	3	0.6	862,976	-0.0001	0.0001	9.30545
-1	1	1	-1	1	-1	-57,529	99,636.5	-287,662	78.6262	-0.0539	3.10183
1	1	-1	1	-1	-1	57,529	-99,636.5	287,655	39.2665	-68.1194	-3.10183
1	-1	1	1	-1	1	115,056	0.6	287,662	-78.6262	0.0539	3.10176

- 600 2.  $z$  and  $\theta_x$  are mainly affected by  $T1D2$ ,  $T2D2$ , and
- 601  $T3D2$ .
- 602 3.  $y$  is mainly affected by  $T2D1$  and  $T3D1$ .
- 603 4.  $\theta_y$  is mainly affected by  $T2D2$  and  $T3D2$ .

604 Also, it is evident that to obtain in-plane displacements  
 605 ( $x$ ,  $y$ ,  $\theta_z$ ), actuators should act in direction one (D1) and  
 606 out-of-plane displacements ( $z$ ,  $\theta_x$ ,  $\theta_y$ ) are generated when  
 607 actuators act in direction two (D2).  
 608

609 4.4 Space filling DOCE and meta-model  
 610 of the HexFlex

611 The preliminary assumption of negligible second-order  
 612 effects was made in order to screen factors on the Plackett–  
 613 Burman fractional factorial design. Because such assump-  
 614 tion had to be verified, a surface response design needs to  
 615 be implemented to confirm the neglect of the second order  
 616 effects assumed in the PB DOCE. Amongst all choices  
 617 available, the uniform design was chosen because the  
 618 physics of the problem at hand prevented using center  
 619 points required by other response surface DOEs such as the  
 620 central composite design. In other words, the center point  
 621 would correspond to the relaxed configuration of the  
 622 mechanism where no movement is achieved.

623 To generate a valid meta-model of the HexFlex, a uni-  
 624 form DOCE [41] was used with the same six factors shown  
 625 in Table 1 and six evenly distributed levels (i.e., -1,  
 626 -0.6, -0.2, 0.2, 0.6, 1). A uniform design is a modifica-  
 627 tion of fractional factorial designs that provides scatter  
 628 design points in the experimental domain space. The design  
 629 matrix and the FEA output displacements were calculated  
 630 using ANSYS<sup>®</sup> (Table 4).

4.5 HexFlex mechanism meta-modeling

632 After running the space filling DOCE (Sect. 4.4), the next  
 633 step consists of choosing an appropriate approximation  
 634 model. Low-order polynomials have been used effectively  
 635 for building approximations in a variety of applications  
 636 including force–displacement modeling [42]. Here a sec-  
 637 ond-order polynomial with interactions is used for meta-  
 638 modeling an input–output of the HexFlex.

639 The chosen polynomial model for the input–output  
 640 meta-model of the HexFlex is shown on Eq. (4).

$$r_i = \beta_0 + \sum_{i=1}^k \beta_i \tau_i + \sum_{i=1}^k \beta_{ii} \tau_i^2 + \sum_{i=1}^k \sum_{j=1}^k \beta_{ij} \tau_i \tau_j \quad (4)$$

642 where  $i = j = 1, \dots, 6$ ,  $i < j$ ,  $\tau_i$ , and  $r_i$  are components of  $\tau$   
 643 and  $\mathbf{r}$ , respectively, as defined in Eqs. (2) and (3).

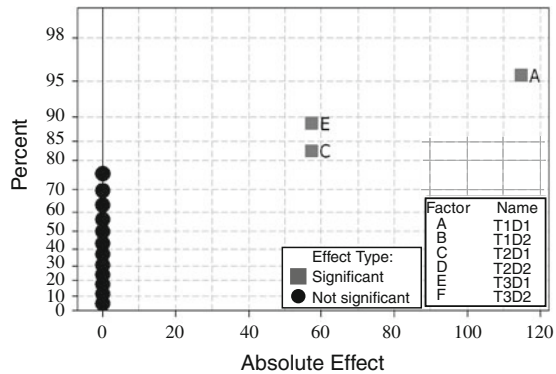
644 The Placket–Burman DOCE analysis points out which  
 645 interactions of the behavior of the mechanism are not  
 646 important and are therefore negligible in Eq. (4). Also, from  
 647 preliminary experiments it is determined that the non-linear  
 648 terms of Eq. (4) do not influence the end effector motion and  
 649 are neglected. As consequence, the meta-model results in a  
 650 system of six linear equations for the  $x$ ,  $y$ ,  $z$ , etc. motions.  
 651 We obtained the force–displacement meta-model writing  
 652 the system of linear equations as a matrix equation (Eq. 5).

$$\begin{bmatrix} x & y & z & \theta_x & \theta_y & \theta_z \end{bmatrix}^T = S_T \begin{bmatrix} T1D1 & T1D2 & T2D1 & T2D2 & T3D1 & T3D2 \end{bmatrix}^T \quad (5)$$

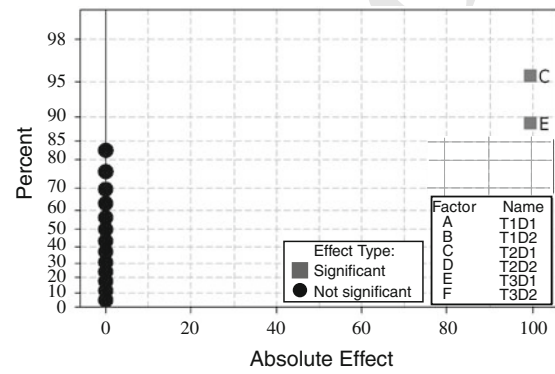
654 where  $S_T$  (Eq. 6) is the matrix representing the input–output  
 655 first-order effects. Each term of  $S_T$  is found using a least squares  
 656 regression [43, 44]. The units associated to the elements  $S_T$  for  
 657 the HexFlex are  $\mu\text{m}/\text{N}$  for rows 1–3, and  $\mu\text{rad}/\text{N}$  for rows 4–6.

Author Proof

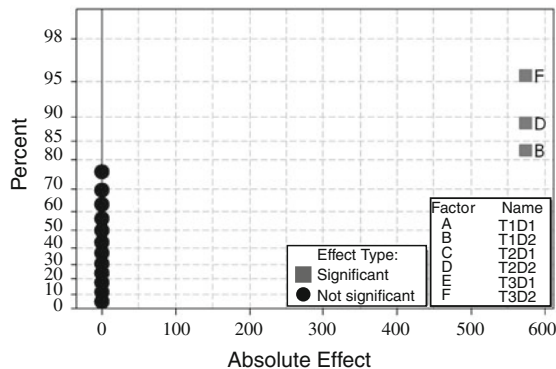
$$S_T = \begin{bmatrix} 57.5 & 0 & 28.8 & 0 & -28.8 & 0 \\ 0 & 0 & 49.8 & 0 & 49.8 & 0 \\ 0 & 287.7 & 0 & 287.7 & 0 & 287.7 \\ 0 & 39,313.0 & 0 & -19,679.9 & 0 & -19,633.3 \\ 0 & 0 & 0 & -34,032.0 & 0 & 34,060.2 \\ -3,101.8 & 0 & 3,101.8 & 0 & -3,101.8 & 0 \end{bmatrix} \quad (6)$$



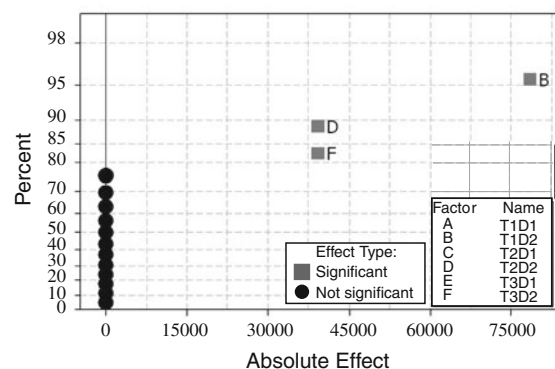
(a) Half Normal Plot for  $x$  translation



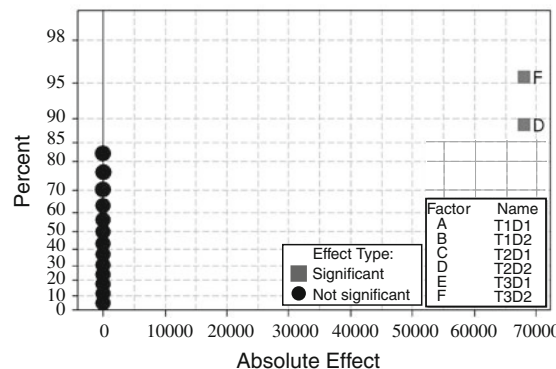
(b) Half Normal Plot for  $y$  translation



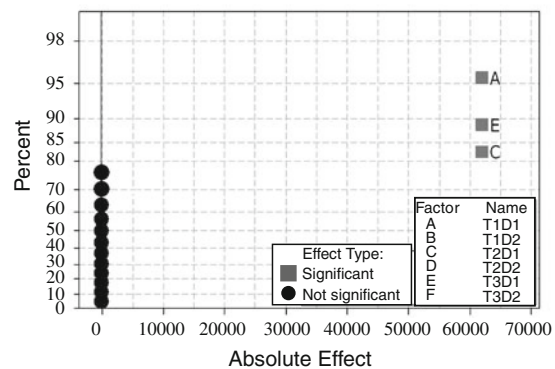
(c) Half Normal Plot for  $z$  translation



(d) Half Normal Plot for  $\theta_x$  translation



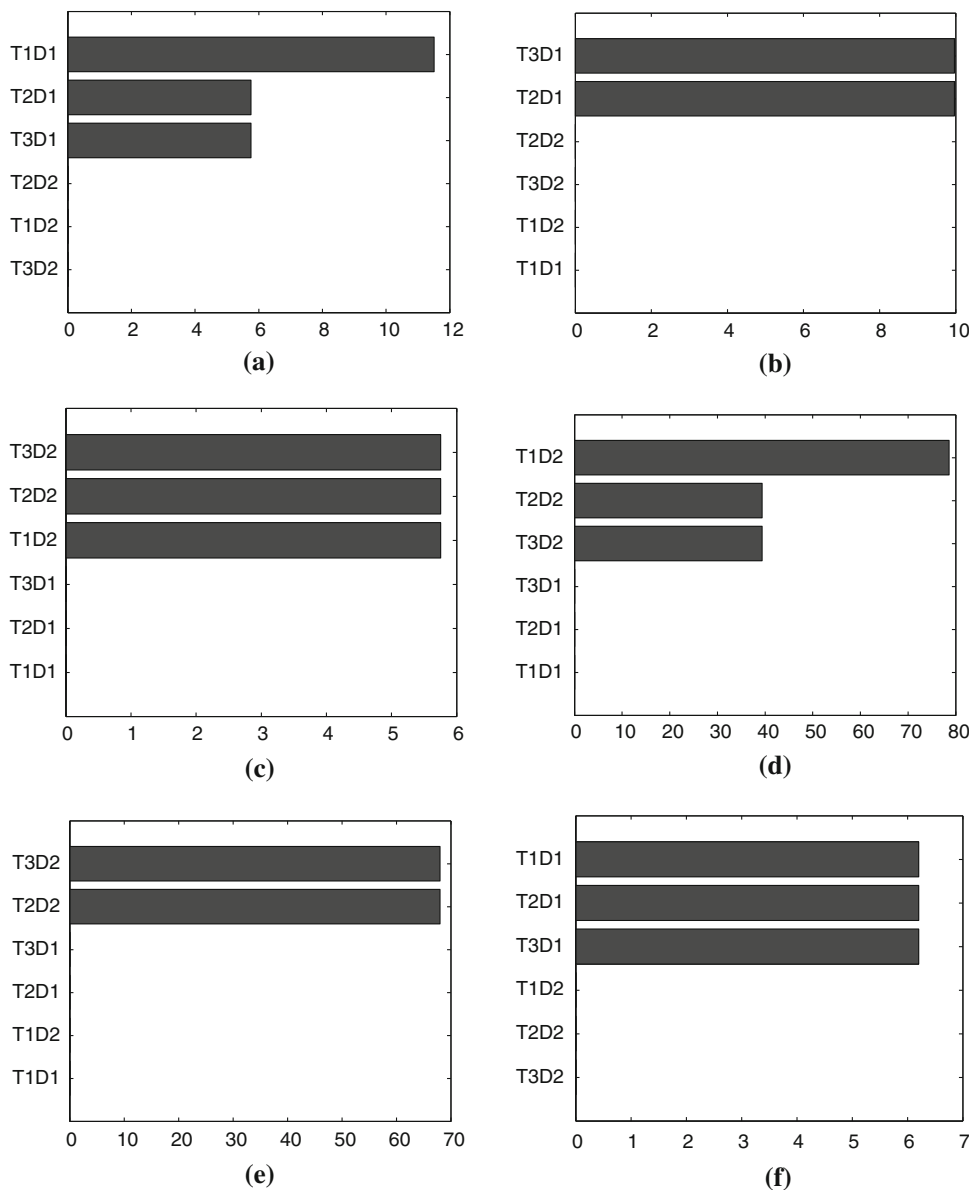
(e) Half Normal Plot for  $\theta_y$  translation



(f) Half Normal Plot for  $\theta_z$  translation

Fig. 7 Half normal probability plots. Plackett–Burman DOCE for 12 runs and 6 factors for HexFlex quasi-static conditions

**Fig. 8** Placket–Burman DOCE for 12 runs and 6 factors for HexFlex quasi-static conditions. **a** Pareto coefficients for  $x$  translation ( $\times 10^4$ ). **b** Pareto coefficients for  $y$  translation ( $\times 10^4$ ). **c** Pareto coefficients for  $z$  translation ( $\times 10^5$ ). **d** Pareto coefficients for  $\theta_x$  rotation. **e** Pareto coefficients for  $\theta_y$  rotation. **f** Pareto coefficients for  $\theta_z$  rotation



Author Proof

658 The scalability and symmetry of the discussed method  
 659 deserve the following comments: (a) a scaled copy of the  
 660 mechanism does not have the scaled I/O function of the  
 661 original mechanism. A reason for this behavior is that size  
 662 has effects on the flexibility of the material. (b) A scaled  
 663 copy of the mechanism accepts the same DOCE method-  
 664 ology proposed, to reach its I/O function. (c) The DOCE  
 665 method applied here does not make use of symmetries of  
 666 the modeled mechanism. However, we did use symmetries  
 667 in checking the values of the obtained coefficients (Eqs. 5, 6)  
 668 with the purpose of finding possible modeling or calcula-  
 669 tion errors. Symmetries helped us to find and correct such  
 670 procedural errors, reaching a meaningful equation.

Although it was not a goal for this article, future work  
 might include the consideration of mechanism symmetries  
 to find the coefficients (not only to check them).

#### 4.6 Validation of the HexFlex meta-model

The resulting residuals patterns (i.e., error vs. predicted  
 values) were found to be aleatory and normally distributed  
 around zero leading to confirm that the error is additive.  
 To validate the accuracy of the meta-model, 1,000 random  
 experiments with uniform distributions and factor levels  
 between  $-1$  and  $1$  N are made. The resulting forward  
 model is used to compare the estimations of the mechanism

**Table 3** Lenth's analysis of six DOF HexFlex mechanism

	$x$ ( $\mu\text{m}$ )	$y$ ( $\mu\text{m}$ )	$z$ ( $\mu\text{m}$ )	$\theta_x$ ( $\mu\text{rad}$ )	$\theta_y$ ( $\mu\text{rad}$ )	$\theta_z$ ( $\mu\text{rad}$ )
T1D2	0.17	0	575,314	78.63	0.05	0
T2D2	0.17	0	575,317	39.36	68.07	0
T3D2	0.17	0	575,321	39.27	68.12	0
T2D1	57,527.50	99,637.1	0	0	0	6.20
T3D1	57,531.50	99,635.9	0	0	0	6.2
T1D1	115,053.50	0	0	0	0	6.20
$v$	19.63	0	287,657	19.63	0.03	3.10
$S_0$	29.45	0	431,486	29.45	0.04	4.70
ME	110.73	0	0	0	0	0
SME	265.34	0	0	0	0	0

**Table 4** Uniform DOCE and results of the experiments

Design matrix							Responses					
Treatment	T1D1 (N)	T1D2 (N)	T2D1 (N)	T2D2 (N)	T3D1 (N)	T3D2 (N)	$x$ ( $\mu\text{m}$ )	$y$ ( $\mu\text{m}$ )	$z$ ( $\mu\text{m}$ )	$\theta_x$ ( $\mu\text{rad}$ )	$\theta_y$ ( $\mu\text{rad}$ )	$\theta_z$ ( $\mu\text{rad}$ )
1	0.6	-0.2	-1.0	-1.0	1.0	-0.2	-23.00	0.00	-403.37	15,723.33	27,294.40	-8,057.19
2	-0.6	-0.6	0.2	1.0	-1.0	0.2	0.00	-39.85	172.88	-47,310.09	-27,294.38	5,583.32
3	-0.2	0.2	-0.2	-1.0	-1.0	0.6	11.5	-59.78	-57.63	15,769.99	54,588.86	3,098.05
4	-1.0	-0.2	1.0	0.6	-0.2	1.0	-23.01	39.85	403.37	-39,390.09	13,647.29	6,831.66
5	1.0	0.2	1.0	1.0	0.6	0.2	69.02	79.71	403.37	-15,723.33	-27,294.40	-1,883.93
6	-1.0	-1.0	-0.2	-1.0	-0.2	-1.0	-57.52	-19.93	-864.35	-116.71	-0.06	3,120.86
7	-0.6	0.2	-1.0	1.0	0.2	0.6	-69.02	-39.85	518.61	-23,596.69	-13,647.16	-1,838.28
8	0.6	-0.6	0.6	0.6	0.2	-1.0	46.01	39.85	-288.11	-15,816.67	-54,588.89	-635.59
9	-1.0	1.0	-0.6	-0.2	-0.6	0.2	-57.52	-59.78	288.10	39,483.43	13,647.20	3,120.87
10	0.2	1.0	0.2	1.0	-0.2	-0.6	23.01	0.00	403.35	31,610.14	-54,588.88	612.76
11	-0.6	0.6	0.2	-1.0	0.2	1.0	-34.51	19.93	172.86	23,690.01	68,236.09	1,872.51
12	1.0	-1.0	-0.2	0.6	-0.6	-0.2	69.02	-39.85	-172.86	-47,356.77	-27,294.41	-1,883.92
13	-1.0	0.6	-0.2	0.6	0.6	-0.2	-80.53	19.93	288.11	15,816.74	-27,294.44	646.99
14	-1.0	0.2	0.6	-0.6	1.0	-0.6	-69.02	79.71	-288.12	31,516.73	-0.05	1,883.92
15	1.0	1.0	0.6	-1.0	-0.2	0.2	80.52	19.93	57.61	55,230.10	40,941.61	-646.99
16	1.0	0.6	-1.0	0.2	0.2	-0.6	23.01	-39.85	57.61	31,563.44	-27,294.47	-6,831.66
17	0.2	1.0	-0.6	-0.6	0.6	-1.0	-23.01	0.00	-172.89	70,976.84	-13,647.30	-4,334.97
18	0.6	0.6	-0.6	0.6	-1.0	1.0	46.02	-79.71	633.86	-7,803.34	13,647.27	-635.57
19	0.2	0.2	0.6	0.2	-0.6	-0.2	46.01	0.00	57.62	7,896.70	-13,647.22	3,086.63
20	0.6	-0.2	0.2	-0.6	-1.0	-0.6	69.02	-39.85	-403.37	15,723.35	-0.04	1,838.29
21	0.6	-0.6	1.0	-0.6	-0.6	0.6	80.52	19.93	-172.86	-23,690.10	40,941.67	3,075.22
22	0.2	-1.0	0.2	0.2	1.0	1.0	-11.50	59.78	57.64	-63,103.51	27,294.51	-3,098.05
23	-0.2	0.6	1.0	-0.2	-1.0	-1.0	46.01	0.00	-172.88	47,310.13	-27,294.50	6,808.84
24	-0.6	-1.0	1.0	-0.2	0.6	-0.2	-23.01	79.71	-403.35	-31,610.10	0.00	3,109.44
25	-0.2	-0.6	-0.6	-0.6	0.2	0.2	-34.51	-19.93	-288.11	-15,816.74	27,294.44	-1,849.69
26	-0.2	-0.6	-0.6	1.0	1.0	-0.6	-57.52	19.93	-57.62	-31,563.37	-54,588.85	-4,323.57
27	-0.6	-0.2	-1.0	0.2	-0.6	-1.0	-46.01	-79.71	-288.12	7,850.04	-40,941.69	635.59
28	-0.2	1.0	0.6	0.2	1.0	0.6	-23.01	79.71	518.60	23,736.74	13,647.23	-612.77
29	1.0	-0.2	-0.2	-0.2	0.6	1.0	34.51	19.93	172.88	-23,643.42	40,941.70	-5,594.73
30	0.2	-1.0	-1.0	-0.2	-0.2	0.6	-11.50	-59.78	-172.85	-47,356.81	27,294.48	-3,098.03

682 configuration using meta-modeling against the FEA soft-  
 683 ware ANSYS<sup>®</sup>. The precision of the model is calculated  
 684 using three error criteria:

- 685 1. The maximum absolute error (MAXABS Eq. (7)).
- 686 2. The relative error between the meta-model and the  
 687 FEA model.
- 688 3. The root mean square error (RMSE Eq. (8)) over the  
 689 set of experiments.

690 The MAXABS and relative % of error allow to calculate  
 691 the local error. The RMSE provides good estimate of the  
 692 global error. The error between meta-model predictions  
 693 and ANSYS<sup>®</sup> results is shown in Table 5. The deformed  
 694 shape of the mechanism for one of the experiments made to  
 695 validate the accuracy of the meta-model is in Fig. 3f.

$$\text{MAXABS} = \max\{|\psi_i - \widehat{\psi}_i|\}_{i=1, \dots, n_{\text{error}}} \quad (7)$$

$$\text{RMSE} = \sqrt{\frac{\sum_{i=1}^{n_{\text{error}}} (\psi_i - \widehat{\psi}_i)^2}{n_{\text{error}}}} \quad (8)$$

699 where  $\psi_i$  refers to  $x$ ,  $y$ , and  $z$  or the corresponding angles  
 700  $\theta_x$ ,  $\theta_y$ , and  $\theta_z$  as shown in Table 5. The maximum linear  
 701 absolute error is in the  $y$  direction ( $4.18 \times 10^{-4} \mu\text{m}$ ). We  
 702 compare this error with an ISO h6 manufacturing tolerance  
 703 calculated on a shaft of nominal diameter 50 mm (50 mm  
 704  $^{+0}_{-19} \mu\text{m}$ ) finding the error as acceptable. The maximum  
 705 relative error is in the  $y$  direction (0.621 %). This repre-  
 706 sents a better accuracy than that reported in Ref. [45] where  
 707 the HexFlex is analyzed using a virtual method based on  
 708 Euler beam theory and the results are also compared with a  
 709 FEA software, obtaining a maximum relative error of 3 %.

## 710 5 Discussion

### 711 5.1 Applicability to other mechanisms

712 Design of experiments, and DOCE, in particular, are  
 713 methods that can be applied to complex systems whose I/O  
 714 function is fully or partially unknown. A mechanism is not

**Table 5** Error between meta-model estimations and ANSYS<sup>®</sup> sim-  
 ulations for 1,000 random experiments with uniform distribution

	MAXABS	MAX (%error)	RMSE
$x (\mu\text{m})$	$4.01 \times 10^{-4}$	$1.08 \times 10^{-3}$	$8.67 \times 10^{-5}$
$y (\mu\text{m})$	$4.18 \times 10^{-4}$	$3.59 \times 10^{-4}$	$2.07 \times 10^{-5}$
$z (\mu\text{m})$	$2.85 \times 10^{-4}$	$2.41 \times 10^{-4}$	$4.19 \times 10^{-5}$
$\theta_x (\mu\text{rad})$	$2.41 \times 10^{-2}$	$2.26 \times 10^{-3}$	$9.78 \times 10^{-4}$
$\theta_y (\mu\text{rad})$	4.16	$6.21 \times 10^{-1}$	$5.57 \times 10^{-3}$
$\theta_z (\mu\text{rad})$	$2.10 \times 10^{-2}$	$2.30 \times 10^{-3}$	$2.23 \times 10^{-3}$

different, in this sense, to a chemical or a biotechnological  
 process. DOCE is able to model other mechanisms, even if  
 they are not of the HexFlex types. The numerical models  
 obtained for different mechanisms will indeed be different  
 in the classification of input and output variables, sensi-  
 tivities, etc. However, the DOCE will still be valid in  
 finding the I/O function of such mechanisms.

### 5.2 Future work

The use of DOCE to yield a transfer function for real time  
 control of CMs has been addressed without considering  
 non-ideal conditions relevant to actual working conditions  
 of the mechanism. We could number a few of these non-  
 idealities: (1) deformation due to wear which would lead to  
 significant changes in dimensions, (2) shape variations due  
 to temperature changes and (3) hysteresis, or material  
 memory. We would expect the initially obtained transfer  
 function by DOCE to be adjusted during the operation of  
 the mechanism but we do not envision using neural net-  
 works to obtain CMs transfer functions because DOCE is  
 more efficient and less time consuming.

Due to the symmetrical shape of the HexFlex CM and  
 that it functions at a very narrow vicinity around the  
 operation point, the resulting low-order polynomial linear  
 meta-models obtained seem realistic and for that reason we  
 consider that it should be interesting to test other compliant  
 mechanisms with no symmetry to see whether linear  
 regression models are adequate or not statistically and to  
 consider using Kriging meta-models for larger experi-  
 mental areas [39]. In addition, it would also be worthwhile  
 to model well-known mechanisms (not necessarily CMs) to  
 verify that their already known transfer functions also  
 result from DOCE.

## 6 Conclusions

This article presents a computer-based meta-model for  
 force–displacement transfer function of CMs under quasi-  
 static conditions using DOCE. A case study is discussed in  
 the domain of a six degrees of freedom HexFlex CM. To  
 obtain the meta-model of the HexFlex, computer experi-  
 ments based on Plackett–Burman and uniform DOCE are  
 performed using FEA. The obtained meta-model of the  
 HexFlex is linear for the movement range of the mecha-  
 nism. The accuracy of the meta-model was calculated by  
 running 1,000 FEA-based computer experiments. The  
 values found in the experiment were compared against  
 those generated by the meta-model. The results of the  
 comparison can be observed in Table 5. They allow to  
 conclude that the chosen meta-model is consistent around  
 the operating point.

763 The factorial DOCE permitted to identify/confirm  
764 characteristics of the mechanism, such as the presence of  
765 symmetries in the actuation and the quasi-static behavior of  
766 the mechanism. A uniform DOCE was employed to fine  
767 tune the model of the mechanism. The mechanism was  
768 modeled using a low-order polynomial, because of its  
769 quasi-static behavior and small displacements. The result-  
770 ing I/O model of the mechanism allows having a transfer  
771 function for developing real-time control. It should be  
772 noted that the (linear) model obtained is easily invertible,  
773 which adds to its applicability. However even if the model  
774 obtained was not linear, its invertibility is guaranteed  
775 because it is a polynomial approximation around the  
776 vicinity of the operating point.

777 Nonetheless, DOCE is not intended to replace real  
778 experiment-based DOE but to forecast/ignore possible  
779 interactions and to fine-tune ranges, therefore, reducing  
780 costs of experimentation and model/product development.

781 **Acknowledgments** Support for this work was provided by the  
782 Colombian Administrative Department of Sciences, Technology and  
783 Innovation (COLCIENCIAS) and the Colombian National Service of  
784 Learning (SENA) grant No. 1216-479-22001. The authors gratefully  
785 acknowledge this support.

## 786 References

- 787 1. Luo Z, Tong L, Wang M, Wang S (2007) Shape and topology  
788 optimization of compliant mechanisms using a parameterization  
789 level set method. *J Comput Phys* 227(1):680–705
- 790 2. Jensen B, Howell L (2002) The modeling of cross-axis flexural  
791 pivots. *Mech Mach Theory* 37(5):461–476
- 792 3. Howell L (2001) *Compliant mechanisms*. Wiley, New York
- 793 4. Lan C, Lee K (2006) Generalized shooting method for analyzing  
794 compliant mechanisms with curved members. *J Mech Design*  
795 128(4):765–775
- 796 5. Venanzi S, Giesen P, Parenti-Castelli V (2005) A novel technique  
797 for position analysis of planar compliant mechanisms. *Mech*  
798 *Mach Theory* 40(11):1224–1239
- 799 6. Booker A, Dennis JE, Frank P, Serafini D, Torczon V, Trosset W  
800 (1997) Optimization using surrogate objectives on a helicopter  
801 test example. In: *Computational methods in optimal design and*  
802 *control*. Birkhäuser, Boston, pp 49–58
- 803 7. Gu L (2001) A comparison of polynomial based regression  
804 models in vehicle safety analysis. In: *ASME 2001 design engi-*  
805 *neering technical conferences*. DETC2001, Pittsburgh
- 806 8. Simpson T, Booker A, Ghosh D, Giunta A, Koch P, Yang R  
807 (2004) Approximation methods in multidisciplinary analysis and  
808 optimization: a panel discussion. *Struct Multidiscip O* 27(5):  
809 302–313
- 810 9. Kodiyalam S, Yang R, LEI G (2004) High-performance com-  
811 puting and surrogate modeling for rapid visualization with mul-  
812 tidisciplinary optimization. *AIAA J* 42(11):2347–2354
- 813 10. Simpson T, Poplinski J, Koch P, Allen J (2001) Metamodels for  
814 computer-based engineering design: survey and recommenda-  
815 tions. *Eng Comput* 17(2):129–150
- 816 11. Crary S (2002) Design of computer experiments for metamodel  
817 generation. *Analog Integr Circ S* 32(1):7–16
12. Culpepper M, Anderson G, Petri P (2002) Hexflex: A planar  
818 mechanism for six-axis manipulation and alignment. In: *ASPE*  
819 *Annual Meeting*, ASPE
13. Culpepper M (2007) Multiple degree of freedom compliant  
820 mechanism. United States Patent, Number: US007270319B
14. Kim C, Moon Y, Kota S (2008) A building block approach to the  
821 conceptual synthesis of compliant mechanisms utilizing compli-  
822 ance and stiffness ellipsoids. *J Mech Design* 130(2):2308–1–11
15. Morsch F, Tolou N, Herder J (2009) Comparison of methods for  
823 large deflection analysis of a cantilever beam under free end point  
824 load cases. In: *ASME 2009 design engineering technical con-*  
825 *ferences*. DETC2009, San Diego
16. Trease B, Moon Y, Kota S (2005) Design of large-displacement  
826 compliant joints. *J Mech Design* 127(4):788–798
17. Dibiaso C, Culpepper M, Panas R, Howell L, Magleby S (2008)  
827 Comparison of molecular simulation and pseudo-rigid-body  
828 model predictions for a carbon nanotube based compliant paral-  
829 lel-guiding mechanism. *J Mech Design* 130(04):2302–1–8
18. Chang S, Lee J, Yen H (2005) Kinematic and compliance anal-  
830 ysis for tendon-driven robotic mechanisms with flexible tendons.  
831 *Mech Mach Theory* 40(6):728–739
19. Yong Y, Lu T (2009) Kinetostatic modeling of 3-RRR compliant  
832 micro-motion stages with flexure hinges. *Mech Mach Theory*  
833 44(6):1156–1175
20. Lobontiu N, Paine J, Garcia E, Goldfarb M (2002) Design of  
834 symmetric conic-section flexure hinges based on closed-form  
835 compliance equations. *Mech Mach Theory* 37(5):477–498
21. Lobontiu N, Paine J, Garcia E, Goldfarb M (2001) Corner-  
836 filleted flexure hinges. *J Mech Design* 123(3):346–352
22. Dado M (2005) Limit position synthesis and analysis of com-  
837 pliant 4-bar mechanisms with specified energy levels using var-  
838 iable parametric pseudo-rigid-body model. *Mech Mach Theory*  
839 40(8):977–992
23. Park SR, Yang ST (2005) A mathematical approach for analyzing  
840 ultra precision positioning system with compliant mechanism.  
841 *J Mater Process Tech* 164–165:1584–1589
24. Banerjee A, Bhattacharya B, Mallik A (2009) Forward and  
842 inverse analyses of smart compliant mechanisms for path gen-  
843 eration. *Mech Mach Theory* 44(2):369–381
25. Boyle C, Howell L, Magleby S, Evans M (2003) Dynamic  
844 modeling of compliant constant-force compression mechanisms.  
845 *Mech Mach Theory* 38(12):1469–1487
26. Tolou N, Herder J (2009) A seminanalytical approach to large  
846 deflections in compliant beams under point load. *Math Probl Eng*  
847 2009:13
27. Zettl B, Szyszkowski W, Zhang W (2005) Accurate low DOF  
848 modeling of a planar compliant mechanism with flexure hinges:  
849 the equivalent beam methodology. *Precis Eng* 29(2):237–245
28. DeBona F, Munteanu M (2005) Optimized flexural hinges for  
850 compliant micromechanisms. *Analog Integr Circ S* 44(2):  
851 163–174
29. Saxena A, Ananthasuresh G (2001) Topology synthesis of com-  
852 pliant mechanisms for nonlinear force-deflection and curved path  
853 specifications. *J Mech Design* 123(1):33–42
30. Zhang S, Fasse E (2001) A finite-element-based method to  
854 determine the spatial stiffness properties of a notch hinge. *J Mech*  
855 *Design* 123(1):141–147
31. Rao CR (1967) Least squares theory using an estimated disper-  
856 sion matrix and its application to measurement of signals. In: *5Th*  
857 *Berkeley symposium on mathematical statistics and probability*,  
858 vol I. University of California Press, Berkeley, pp 355–372
32. Plackett RL, Burman JP (1946) The design of optimal multifac-  
859 torial experiments. *Biometrika* 33 (4):305–325
33. Fang K, Li R (2006) Uniform design for computer experiments  
860 and its optimal properties. *Int J Mater Prod Tec* 25(1–3):198–210

- 883 34. Culpepper M, Anderson G (2004) Design of a low-cost nano- 898  
 884 manipulator which utilizes a monolithic, spatial compliant 899  
 885 mechanism. *Precis Eng* 28(4):469–482 900  
 886 35. Craig J (1989) *Introduction to Robotics*, 2nd edn. Addison Wesley, 901  
 887 USA 902  
 888 36. Paul R (1981) *Robot manipulators: mathematics, programming, 903  
 889 and control*. MIT Press, Cambridge 904  
 890 37. Box G, Hunter J, Hunter W (2005) *Statistics for experimenters: 905  
 891 design, innovation, and discovery*, 2nd edn. Wiley, New York 906  
 892 38. NIST (2008) *Engineering statistics handbook.nist/sematech 907  
 893 e-handbook of statistical methods* 908  
 894 39. Kleijnen J (2008) *Design and analysis of simulation experiments.* 909  
 895 Springer, Heidelberg 910  
 896 40. Lenth RV (1989) Quick and easy analysis of unreplicated fac-  
 897 torials. *Technometrics* 31(4):469–473
41. Fang KT, Lin DK, Winker P, Zhang Y (2000) Uniform design: 898  
 theory and application. *Technometrics* 42(3):237–248 899  
 42. Simpson TW, Peplinski J, Koch PN, Allen JK (1997) On the use 900  
 of statistics in design and the implications for deterministic 901  
 computer experiments. In: *ASME 1997 design engineering 902  
 technical conferences, DETC1997* 903  
 43. Bates DM, DebRoy S (2004) Linear mixed models and penalized 904  
 least squares. *J Multivariate Anal* 91(1):1–17 905  
 44. Ruppert D, Wand MP (1994) Multivariate locally weighted least 906  
 squares regression. *Ann Stat* 22(3):1346–1370 907  
 45. Petri P (2002) *A continuum mechanic design aid for non-planar 908  
 compliant mechanisms*. M.Sc thesis, MIT press Cambridge 909  
 910

UNCORRECTED PROOF